SM-29E SUPER FULCRUM

A Transformational Upgrade For MiG-29, MIG-29S/SE/SM/SMT and MiG-29M Tactical Aircraft Engineered, Installed and Supported by Stavatti Aerospace



- Raytheon AN/APG-79 AESA Radar
- Stavatti Design Conformal Fuel Tanks (CFTs)
- Gen III Helmet Mounted Display System (HMDS) or JHMCS
- Sparrowhawk HUD or BAE LiteHUD
- L3 580mm x 203mm Large Area Avionics Display (LAAD)
- New Design External Pylons with LAU-129 MRLs
- Increased Internal Fuel Capacity to 7,214 Ltrs
- Increased External Warload to 6,800 kg
- Comprehensive Integrated ECM System
- · Improved Wheels, Tires, Brakes and Anti-Skid
- Optional Retractable In-Flight Refueling Probe



DESCRIPTION

The SM-29E Super Fulcrum is an enhanced performance MiG-29 Fulcrum aircraft that has been transformationally upgraded by Stavatti Aerospace to the "Super Fulcrum" configuration, allowing the aircraft remain a potent and competitive air defense asset through 2040. The proposed upgrades include installation of new GEAE F414 EPE afterburning turbofan engines and re-engineered engine nacelles. The F414 EPE engines will provide greater thrust, lower specific fuel consumption and improve maintainability over the original RD-33 powerplants. The modifications include sensor, avionics and cockpit upgrades including installation of a Raytheon APG-79 AESA radar, Stavatti proprietary Conformal Fuel Tanks (CFTs), a L3Harris Large Area Avionics Display (LAAD), a comprehensive internal ECM/EW suite, an upgraded HUD and advanced Hands on Throttle and Stick (HOTAS). Proposed modifications include NATO standard communications, navigation and electronic warfare systems as well as new wheels, tires, brakes and antiskid. For enhanced mission capability, newly designed pylons and ejectors are proposed to provide for delivery of both current Ukraine Air Force weapons and missiles and NATO ordnance including the AIM-9, AIM-120, JDAM and JSOW weapons. Major airframe components are to be inspected and rebuilt as needed. Structures that can benefit from advanced materials are to be replaced to provide for a new total airframe life of 6,000 flight hours. To increase aircraft range the Conformal Fuel Tanks (CFTs) increase aircraft total internal fuel capacity to 7,214 Ltrs. A retractable in-flight aerial refueling probe is also available as a cost plus option.

SPECIFICATIONS

Aircraft: SM-29E Super Fulcrum Original Manufacturer: MiG

Upgrade Of: MiG-29, MiG-29S/SE/SM/M/SMT/G Upgrade Provider: Stavatti Aerospace Ltd

Accommodation

Crew SM-29E Single Seat; SM-29F Two Seat Seating K-36DM Ejection Seat

Armament

Internal Gsh-301 30mm cannon with 150 rds External 7 External Stores Hardpoints

Powerplant

2
Afterburning Turbofan
GEAE
F414 EPE
11,938 kgf st
16,800 kgf st
23,876 kgf st

Dimensions

Max Wingspan	11.36 m
Max Length	16.28 m
Max Height	4.73 m
Wing Area	38 sq m

Weights

Empty Operating Max Internal Fuel Max Internal + External Fuel Max Warload Typical Takeoff Weight (TTW) Typical Combat Weight (TCW)	11,330 kg 5,792 kg 8,708 kg 6,800 kg 17,717 kg 14,979 kg

Loadings

Wing Loading-TCW	394 kg/sq m
Thrust/Weight-TCW	1.35 to 1
Wing Loading-MTOW	621 kg/sq m
Thrust/Weight-MTOW	1.01 to 1
Limit Load Factor-TCW	+10.0 g
Limit Load Factor-MTOW	+7.50 g

Avionics & Electronic Warfare

Radar	AN/APG-79 AESA
IRST	IRST21
Databus	MIL-STD-1553-B
HMDS	Gen III HMDS or JHMCS
HUD	Sparrowhawk or LiteHUD
HDD	508mm x 203mm LAAD
Comm	AN/ARC-210(V) Gen V
IFF	AN/APX-126,125(V) or 113(V)
Data Link	TACR-16DL
MMDP	FV-4000
GPS/INS	FALCN
RF ECM	AN/ALQ-211A(V)4 (Optional)
RWR	AN/ALQ-211A(V)4 (Optional)
MAWS	AN/AAR-58 (Optional)
SPJ	AN/ALQ-214 (Optional)
Chaff/Flare	BVP-30-26M & AN/ALE-47

Performance

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Stall Speed, TTW-SL	222 km/hr
Stall Speed, TLW-SL	189 km/hr
Stall Speed, MTOW-SL	256 km/hr
Takeoff Speed, TTW-SL	244 km/hr
Takeoff Speed, MTOW-SL	282 km/hr
Approach Speed, TLW-SL	228 km/hr
Approach Speed, MTOW-SL	287 km/hr
Max Level Speed-SL	1.22 Mach
Max Level Speed-FL360	2.55 Mach
Max ROC, TCW-SL	423.6 m/sec
Service Ceiling	18,898 m
Tactical Radius, Internal Fuel	898 km
Max Range, Internal Fuel	2,393 km
Ferry Range (no inflight refueling)	3,882 km
Take-off Run, TTW	228 m
Landing Run, TLW	601 m





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